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## RESEARCH ARTICLE

10.1029/2018EA000463

## What We Can Learn from the Electric Field and Conductivity Measurements in Auroral Atmosphere

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## Key Points:

- Understanding the mechanisms that control the electric properties of the auroral atmosphere
- Test the instrument performances in a radiation environment close to that we expect to encounter near the Mars surface

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**Abstract** In this paper we present measurements of the electric field and electric conductivity of air performed with the Short Dipole Antenna (SDA) in the auroral atmosphere. The observations were carried out during two stratospheric balloon flights in the northern Sweden in winter of 2011. Two-fold objectives were attained as the outcome of these flights and the data analysis. First aimed at understanding the mechanisms that control the electric properties of the auroral atmosphere and at estimation of some key parameters from the measured data, such as the charge and current density, the mobility of the small-ions, their concentration and generation rate, as well as the radiation dose rate. Second objective was to test the SDA instrument performances in a radiation environment close to that we expect to encounter near the Mars surface.

## 1. Introduction

Earth's atmosphere represents a unique natural laboratory that offers a wide variety of electric conditions and also multiple mechanisms which controls its electrical features. From ground to upper stratosphere the electric conductivity of air increases from  $\sim 10^{-14}$  to  $\sim 10^{-10}$  S m<sup>-1</sup>, while the continuous electric field decreases from few hundreds of V m<sup>-1</sup> down to tens of mV m<sup>-1</sup>. Locally and during short periods of time, the electric field might grow up to  $\sim 100$  kV m<sup>-1</sup>. Such fields are observed during thunderstorms in the troposphere or during dust activation phenomena in deserts. In spite of such a variety, it is difficult to find a natural environment that imitates at once the main electric parameters that we expect to encounter near the Mars surface (see, for example, Molina-Cuberos et al., 2006; Farrell & Desch, 2001). If we are looking for a high electric field generated by dust electric charging, we shall explore hot and arid desert areas (see, for example, Metzger, 1999; Balme & Greeley, 2006; Metzger et al., 2011; Seran et al., 2013). The electric charge carried by the surface and the airborne particles depends on the transport mechanisms, strength of winds, pressure and temperature gradients, dielectric proprieties of the soil material, amount of available dust on the soil surface. If we are looking for relatively high electric conductivity of air, we shall ascend to the upper atmosphere or dare to venture in an underground cavity hosted in granitic rocks. In the first case, the air ionization is produced by galactic cosmic rays and solar energetic particles (see, for example, Nehar, 1967; Reitz, 1993; Viggiano & Arnold, 1995; Singh et al., 2011; Iles et al., 2004). In the second case, the ions of both polarities are generated by a decay of the <sup>222</sup>Rn gas emitted from the surrounding rocks (see, for example, Richon et al., 2005, Seran et al., 2017). In the first case, the radiation level varies with the solar activity, atmospheric height, geomagnetic latitude and solar zenith angle. In the second case, the <sup>222</sup>Rn gas activity concentration is defined by the connectivity between the rock matrix and fractures leading to the underground cavity, combined with natural ventilation and decay.

In this paper we present the measurements of the electric field and electric conductivity of air performed with the Short Dipole Antenna (SDA) during two night flights carried out in the Northern Sweden in February and March of 2011. These observations are analyzed together with the environmental and geomagnetic parameters, such as the height variation of the air temperature, pressure and wind speed, the near ground magnetic and electric fields, the photo-emissions in the lower ionosphere. Simple model is proposed to characterize the radiation environment of the lower stratosphere and to get the relationship between the radiation rate dose, ion density and electric conductivity of air. Impact of the magnetic and thunder storms on the electric properties of atmosphere is also considered.

The structure of this paper is the following. We start with descriptions of the electric field and optical instruments (Section 2), the balloon flights and the environmental conditions (Section 3). In Section 4 we present

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**Figure 1.** SDA gondola on its sledge placed in the flight chain.

the measurements of the electric field and air electric conductivity versus height. The observations of the ultra-low frequency pulses and the resonances in the extreme-low frequency range, produced by the lightning strokes are analyzed in the same Section. In Section 5 we deduce few parameters from the measured data set, such as mobility of small-ions, their concentration and generation rate. The estimation of the radiation dose rate is given in the sub-Section 5.3. The main findings are summarized in Section 6.

## 2. Instrumentation

### 2.1. SDA Gondola

The SDA gondola (Figure 1) was developed specifically for the balloon flights of the SDA instrument. External structure of the gondola was design to ensure the mechanical and thermal protection of the SDA electronics. The gondola 8 cm-thick walls were made from the polystyrene enforced with fiberglass, were painted with an electrically conducting painting and covered with conducting aluminum plates which are electrically connected to the electronics box. A GPS with pulse per second signal

(PPS) was integrated in the gondola and provided the absolute time of the SDA data and gondola location. A specific sledge was used during the balloon launch phase.

### 2.2. Short Dipole Antenna (SDA)

The SDA instrument consists of 3 cylindrical electrodes mounted on the gondola with insulated masts (Figure 1) and electronics box accommodated inside the gondola. The masts were designed to be reasonably short (~50 cm) to reduce a possibility of their damage during the balloon launch phase and sufficiently long to minimize the electric perturbations induced by the gondola. Each electrode measures the electric potential from DC (direct current) to ~3.2 kHz with respect to the 0 V- electric reference, has a dynamical range of  $\pm 125$  V and sensitivity of few  $\mu$ V. In addition, the measurements of the electric conductivity of air were performed periodically during each flight by sending a quick voltage perturbation at one of the electrodes. Measuring the characteristic time of the potential relaxation to the initial unperturbed level gives an estimation of the coupling resistance between the electrode and the atmosphere. The coupling resistance is a parameter that depends not only on the electric conductivity of the air, but also on the measurement method and on the instrument geometry.

A simplified schematic of the SDA electrical configuration, relationship between the measured potential and that found at the electrode level and also the relationship between the coupling resistance and the air electric conductivity were discussed in details in our previous publications, i.e. Seran et al. [2013 and 2017]. Here, we briefly summarize the main findings.

A The temporal variation of the measured potential  $\phi_1(t)$  during the time relaxation measurements reads as:

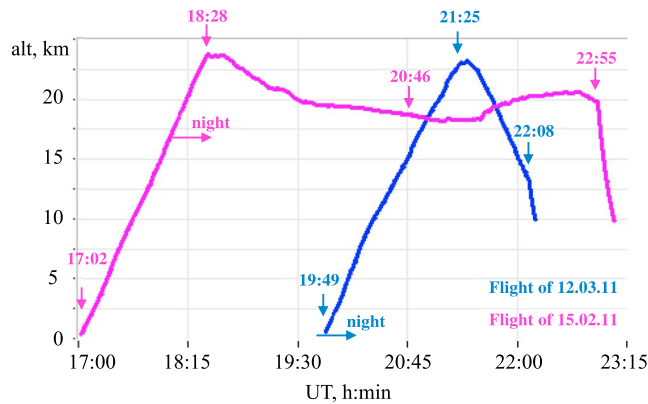
$$\phi_1(t) = \phi_0 \frac{C_0}{C} e^{-\frac{t}{R_2 C}}, \quad (1)$$

here  $\phi_0$  is a positive or negative voltage, which is sent at the preamplifier input through a capacitance ( $C_0$ ) at the initial phase of a time-relaxation sequence. The effective coupling of the electrode to the atmosphere is represented by its resistance ( $R_2$ ), the electrode capacitance by  $C_2$  and the preamplifier by its input capacitance ( $C_1$ ). Thus,  $C = C_0 + C_1 + C_2$ . The preamplifier input resistance is omitted in (1), because its value is significantly higher than the upper limit of the air resistance in the Earth's atmosphere (up to few  $10^{14} \Omega$  near the ground).

B The relationship between the air electric conductivity and the coupling resistance reads as:

$$\sigma = \frac{\epsilon_0}{C_2 R_2}, \quad (2)$$

where  $\epsilon_0$  is the permittivity of vacuum.



**Figure 2.** Altitude profiles of two balloon flights of 15.02.2011 (magenta) and 12.03.2011 (blue).

C The relationship between the measured potential ( $\varphi_1$ ) and the potential at the electrode level ( $\varphi_2$ ) is found to be the following:

$$\phi_1/\phi_2 = (1 + i\omega C_2 R_2)/(1 + i\omega(C_0 + C_1 + C_2)R_2), \quad (3)$$

here  $\omega$  is a circular frequency. This relationship is valid for the floating potential measurements. At the frequencies well below  $[(C_0 + C_1 + C_2)R_2]^{-1}$ , the  $\phi_1/\phi_2 = 1$ , and at the frequencies well above  $[C_2R_2]^{-1}$ , the  $\phi_1/\phi_2 = C_2/(C_0 + C_1 + C_2)$ . Substituting the SDA parameters, i.e.  $C_0 + C_1 \approx 2.1pF$  and  $C_2 \approx 2.9pF$ , the last ratio goes down to  $\sim 0.58$ . In the conditions, when the coupling resistance varies from  $10^{14}$  to  $10^{11} \Omega$ , the upper transient frequency  $[C_2R_2]^{-1}$  moves from  $\sim 3 \text{ ms}^{-1}$  to  $3 \text{ s}^{-1}$ .

The method described above is used henceforth to correct the measured electric potentials, to estimate the coupling resistance and to deduce the air electric conductivity.

### 2.3. Optical Camera ALFA

The ALFA (Auroral Light Fine Analysis) mobile all-sky camera [Seran et al., 2009] detects the photo-emissions in the wavelength range from  $\sim 400$  to  $700 \text{ nm}$ . The acquired images have a spatial resolution of  $150 \text{ m}$  in the zenith and of  $\sim 5 \text{ km}$  in the horizon when assuming photo-emissions at the altitude of  $150 \text{ km}$ . The exposure time was fixed at  $6 \text{ s}$  and the image sensitivity at  $1600 \text{ ISO}$ . The images were taken each  $40 \text{ s}$ . This camera operated during both balloon flights.

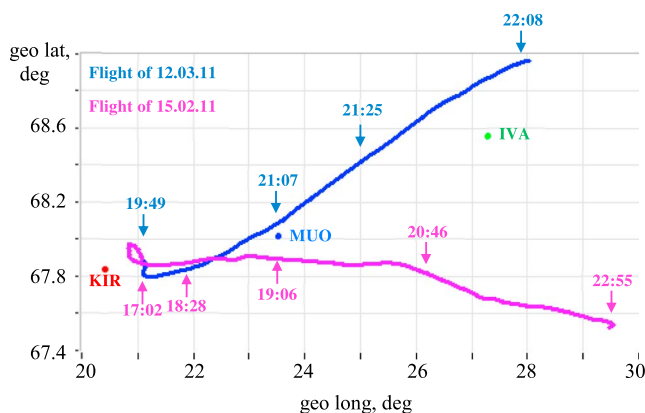
## 3. Balloon Flights and Environmental Conditions

### 3.1. Balloon Flights

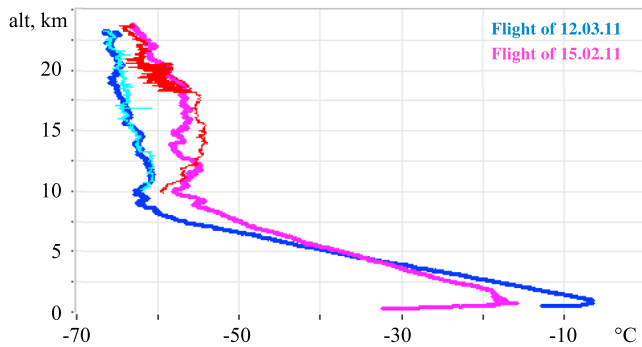
In frame of the CNES (Centre National d'Etudes Spatiales) balloon campaign in winter 2011, a number of 35SF- open stratospheric balloons was launched from the ESRANGE European Space Center located in northern Sweden ( $67^\circ 53'$  geo lat and  $21^\circ 5'$  geo long). The SDA gondola was a part of scientific payloads for two balloon flights, i.e. on February, 15 and on March, 12. In both cases, the balloons were released after the sunset (or more precisely, after the nautical dusk for the first flight and after the astronomical dusk for the second flight), reached their ceilings at  $\sim 23\text{--}24 \text{ km}$  in around  $1 \text{ h } 30 \text{ min}$  (Figure 2), drifted eastward and traveled more than  $300 \text{ km}$  (Figure 3). In contrast with the flight of February, 15, when the balloon was maintained at an altitude of  $\sim 20 \text{ km}$  during about 4 hours, the phase of “slow” descent on March, 12 started immediately after the end of the balloon ascent phase and lasted  $\sim 43 \text{ minutes}$ . Final stages of the descent after the balloon release were carried with a parachute.

### 3.2. Pressure and Temperature of Air Versus Height

A monitoring of the air pressure and temperature during each flight was performed with the CNES equipment accommodated in the balloon payload. Temperature profiles for two balloon flights measured during ascent and descent phases are shown in Figure 4. An abrupt change in the temperature gradient is clearly observed at two different heights. Near the ground, this change corresponds to the inversion layer, i.e. the layer situated between the ground and  $\sim 1 \text{ km}$ , where the warmer air is found above the colder one. This layer is typical for the polar regions during winter, when the radiation from the earth surface exceeds the solar radiation. Another change in the temperature gradient is observed at the height of  $\sim 8\text{--}10 \text{ km}$  and corresponds to the tropopause, i.e. the boundary between the troposphere and stratosphere. Below this boundary the temperature decreases with height with a lapse rate of  $\sim 5 \text{ }^\circ\text{C km}^{-1}$  and  $7 \text{ }^\circ\text{C km}^{-1}$ , respectively, on February, 15 and on March, 12. Above, the average lapse rate does not exceed  $\sim 2 \text{ }^\circ\text{C km}^{-1}$ .



**Figure 3.** Trajectories of two balloon flights launched from ESRANGE on February, 15 (magenta) and on March, 12 (blue). Positions of three magnetic stations of IMAGE network are indicated by filled circles.



**Figure 4.** Air temperature versus height for two balloon flights measured during ascent (in magenta for 15.02.11 and in blue for 12.03.11) and descent (in red for 15.02.11 and in cyan for 12.03.11) phases.

The profile of the atmospheric pressure measured during the balloon flight in March is identical to that recorded in February. The pressure varies exponentially with altitude with the characteristic heights of 7 and 6.2 km, respectively, below and above the altitude of  $\sim 8$  km. The mass density of the  $N_2$ -dominated atmosphere deduced from the measured temperature and pressure follows the same law as the gas pressure with the characteristic heights of 9 and 6.2 km, respectively, below and above the altitude of  $\sim 8$  km. The variation of both parameters with height is displayed in Figure 5.

### 3.3. Geomagnetic and Weather Conditions

The global magnetic activity level was rather quiet during the February, 15 flight and fairly high during March, 12 flight. Shown in Figure 6 are the variations of the north component of the magnetic field recorded by the Muonio (MUO) magnetometer of the northern Fennoscandia IMAGE network, situated close to the footprint of both balloon trajectories (Figure 3). Negative deviation in the north magnetic component all along the March, 12 flight corresponds to the substorm growth phase. Longitudinal distribution of the wide-band photo emissions versus time is displayed at the same time scale as the magnetic field records. This keogram is produced from the ALFA all-sky images (Section 2.3). Each column corresponds to a set of pixels along the line that connects the image center with the image east edge. Thus, assuming that the recorded photo-emissions are mainly generated at the altitude of  $\sim 140$  km, each column of the keogram corresponds to the longitude coverage of the all-sky camera eastward from the ESRANGE.

In contrast with the March, 12, sky was covered with clouds and magnetic activity was low during first two-thirds of the February, 15 flight, but slightly increased after 21 UT. The SDA instrument operated until 20:46 UT during this flight.

## 4. Observations

The method described in the Section 2.2 is used henceforth to correct the measured electric potentials, to calculate the vertical electric field and the air electric conductivity.

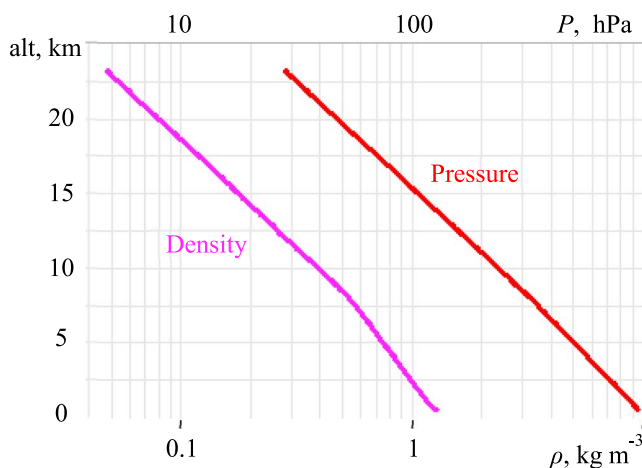
### 4.1. Vertical Component of the Electric Field Versus Height

Vertical component of the electric field,  $E_V$  (in  $V m^{-1}$ ), was measured during both ascent and descent phases of the February, 15 and March, 12 balloon flights. Its variation with height,  $z$  (in km), is shown in Figure 7 and is found to approximately follow the exponential function:

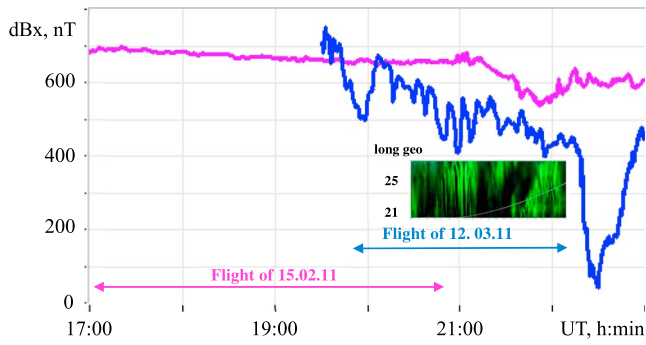
$$E_V(z) = E_0 \cdot \exp(-z/h_0). \quad (4)$$

Both parameters, i.e. the characteristic height,  $h_0$ , and amplitude,  $E_0$ , are height-dependent. Their amplitudes that provide the best fit to the measured series of data points are indicated in the Table 1.

The change in the slope of the vertical electric field observed during both flights is related to the sharp variation of the horizontal wind at the altitudes of 15 and 10 km, respectively, for the February, 15 and March, 12 (see the right panel of Figure 7). The wind shear rate increases at these altitudes from  $\sim 0.7 \cdot 10^{-3} s^{-1}$  to  $\sim 3-4 \cdot 10^{-3} s^{-1}$ . In addition to the slope change, a slight shift in the measured electric field amplitude is seen in the ceiling and at  $\sim 15$  km. This shift is caused by a quick variation of the balloon vertical speed.



**Figure 5.** Measured air pressure (in red) and deduced air mass density (in magenta) versus height. The air pressure is fitted with:  $P(z) \sim 1030e^{-z/7}$  below  $z = 8$  km and  $P(z) \sim 1160e^{-z/6.2}$  above  $z = 8$  km. The air mass density is fitted with:  $\rho(z) \sim 1.3e^{-z/9}$  below  $z = 8$  km and  $\rho(z) \sim 2e^{-z/6.2}$  above  $z = 8$  km.



**Figure 6.** Variations of north component of the magnetic field recorded by MUO magnetometer during balloon flights on February, 15 (in magenta) and on March, 12 (in blue). Longitudinal distribution of light emission versus time deduced from the ALFA all-sky images is also displayed for March, 12 flight. Balloon trajectory is depicted by white line.

Well above from the ground, the electric field is produced by a misbalance between the ions of opposite polarities. The relationship between the electric field and the space charge density,  $\rho$ , is given by the Poisson equation, which in assumption that the main variations of the electrical parameters in atmosphere are vertical reads as:

$$\frac{dE_V}{dz} = \frac{\rho}{\epsilon_0}. \quad (5)$$

According to (5), the positive space charge produces the positive derivative of the vertical electric field, like that measured during our balloon flights. Substituting the (4) in the (5), the variation of the space charge density versus height reads as:

$$\rho(z) = -\frac{\epsilon_0 E_0}{h_0} \cdot \exp(-z/h_0). \quad (6)$$

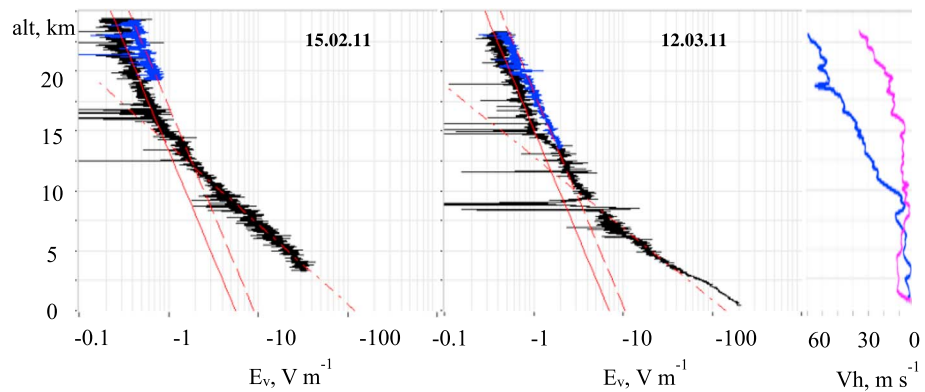
The variation of this parameter with height is shown in Figure 8. Taking into account that the ion concentration in the low stratosphere is of order of  $10^{10} \text{ m}^{-3}$  and that the charge density at these heights is of order of  $2 \cdot 10^{-15} \text{ C m}^{-3}$  (thus, of order of  $10^4 \text{ e m}^{-3}$ ), one may conclude that only very insignificant fraction ( $\sim 10^{-6}$  for singled-charged ions) of ions is not neutralised by the ions of opposite polarity. The discontinuity of the space charge density corresponding to the sharp change of the electric field gradient is found at (or just above) the tropopause. Above this boundary, the number of the misbalanced charges is divided by three.

#### 4.2. Electric Conductivity of Air Versus Height

The electric conductivity of air,  $\sigma$  (in  $\text{S m}^{-1}$ ), is a parameter which characterizes the capacity of light ions to carry the electric current in the atmosphere. This parameter is determined by the concentration of small-ions,  $n_i$  (in  $\text{m}^{-3}$ ), their mobility,  $\mu$  (in  $\text{m}^2 \text{ V}^{-1} \text{ s}^{-1}$ ), and the electric charge carried by each ion,  $q$  (in C). It reads as:

$$\sigma = q\mu n_i. \quad (7)$$

Even if this parameter was measured during entire flight, the most points depicted in the Figure 9 correspond to the observations performed in the stratosphere. The reasons of this are discussed in our previous paper [Seran et al., 2017] and are related to the limitations of the relaxation method. The method works well when the electric field produced by the potential applied to the electrode in the initial phase of the relaxation



**Figure 7.** Vertical component of the electric field versus height, as measured during balloon flights on February, 15 (left) and on March, 12 (centre). Measurements performed during the ascent and descent phases are shown in black and blue, respectively. Red lines correspond to the best fit (4) with the parameters given in Table 1. Horizontal wind speed versus height, measured during February, 15 (mauve) and March 12 (blue) flights, are shown on the right panel.

**Table 1**  
Characteristic Parameters of the Vertical Electric Field Exponential Variation With Height

parameter	February flight			March flight		
	ascent	ascent	descent	ascent	ascent	descent
	$3 < z < 12.5$	$12.5 < z < 24$		$3 < z < 10$	$10 < z < 23$	
$h_0$ , km	2.9	7.8	7.8	2.6	7.8	7.8
$E_0$ , $V m^{-1}$	-120	-5.6	-9	-140	-7	-10.5

measurements is significantly higher than the ambient DC and AC (continuous and alternative) electric field. This constraint does not create any problems for measurements made in the quiet Earth's stratosphere with electric field of a fraction of  $V m^{-1}$ , but it does if the instrument is used in lower atmosphere where the electric field exceeds  $\sim$ few  $V m^{-1}$ . With aim to extend the application domain of the instrument, a new design of the conductivity sensor was recently proposed and validated (see for the details Seran et al., 2017). In this design the sensitive part of the sensor is isolated from the external electric field.

Similar to the electric field, the electric conductivity of air varies exponentially with height, but its amplitude increases with altitude. The measured values are found to match:

$$\sigma_{\pm}(z) = \sigma_0 \cdot \exp(z/h_0), \quad (8)$$

with  $\sigma_0 \approx 2.3 \cdot 10^{-13} S m^{-1}$  and  $h_0 = 7.8$  km. The conductivities of both polarities are estimated to be close to each other, with a minor surplus of  $\sim$ 2% for the negative conductivity. However, this excess is insignificant with respect to the mean absolute deviation of the measured points, which is  $\sim$ 8%.

Careful analysis of the air conductivity recorded during the March, 12 flight indicates that the measured magnitudes are slightly less during balloon ascent than during its descent. This difference might be explained by a change of the auroral activity in time and with respect to the balloon location. During its ascent, the balloon crosses the edges of few thin isolated arcs (left keogram in Figure 9), i.e. eastward from the main region of the auroral emissions, while during its descent, the balloon moves through a diffuse part of the auroral oval (right keogram in the same figure). Of course, the most energy of the energetic electrons precipitated during magnetic storms along the Earth magnetic field lines towards the atmosphere is lost in the ionosphere, i.e. at the altitudes between  $\sim$ 110 and 350 km. This energy goes to the ionisation of the atmospheric atoms and molecules, production of thermal electrons, generation of photo-emissions. However, some part of very energetic electrons (with energies above few tens of keV) can reach the lower altitudes and contribute to the air ionisation. Even if the SDA data seems to give an indication in favour of such mechanism, the additional observations in conditions of stronger auroral activity have to be performed to approve or disapprove these observations.

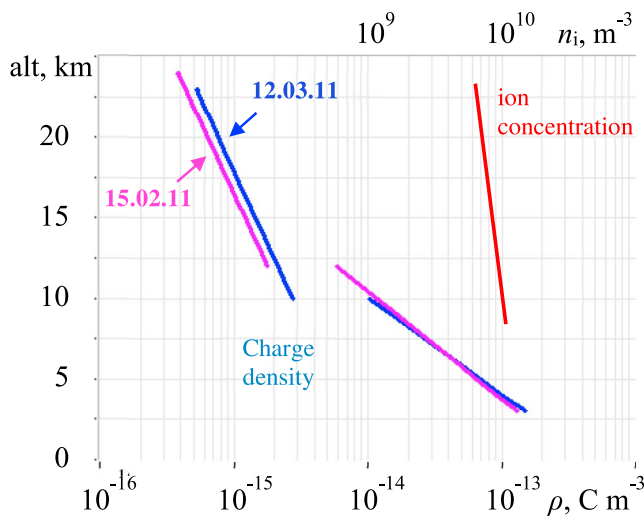
The fact that the electric field and electric conductivity profiles have the same characteristic height indicates that the density of the vertical electric current that circulates in the stratosphere is nearly constant and equals to

$$j_v(z) = (\sigma_+ + \sigma_-)E_v \approx 3-4 pAm^{-2}. \quad (9)$$

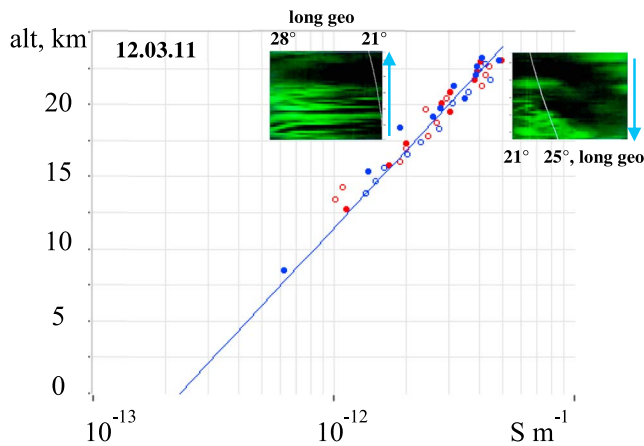
### 4.3. Lightning Detection in Stratosphere

Although in February–March the lightning activity attains its minimum in Europe, the lightning detection networks still record in average around few lightning strokes per minute. These strokes are mainly produced in the southern Europe, i.e. below  $\sim$ 45° geo lat (see, e.g. Anderson & Klugmann, 2014). The highest thunderstorm activity with, in average, one lightning per second is usually observed in July. Of course, during thunderstorm periods, the lightning rate is increased well above the above mentioned average values.

One example of the positive and negative lightning strokes detected simultaneously by SDA and EUCLID network [see, e.g., Schulz et al., 2016] is presented in Figure 10. These two strokes with the peak currents,  $I$ , of +159 kA and  $-67$  kA induce the variation of the vertical electric field,  $E_v$ , of +130 mV



**Figure 8.** Charge density versus height, as deduced from the electric field measured during the balloon flights on February, 15 (magenta) and on March, 12 (blue). Small-ion concentration is shown in red.



**Figure 9.** Electric conductivity of air versus height, as measured during balloon flight on March, 12. Solid and empty circles are used to show the measurements during the ascent and descent phases, respectively. Blue and red colours indicate the negative and positive ions conductivities, respectively. Longitudinal distributions of the wide-band photo emissions versus time are displayed separately for the balloon ascent (on left) and descent (on right) phases. Balloon trajectory is indicated by white line.

and  $-33$  mV at the distance,  $d$ , more than 3000 km from the source region. Comparing the measured amplitudes with those predicted by the transmission line model with  $E_v \sim 20 I/d$  (see, e.g., Uman et al., 1975), one may conclude that only about 10% of the radiation field is transmitted at such distances. The rest is likely absorbed in the atmosphere or contributes to the excitation of waves in the extreme-low frequency range (see, e.g., Price, 2016). With aim to demonstrate the relationship between the lightning intensity and the Schumann resonances magnitudes, their temporal variations are shown in Figure 11 for the time period of 10 minutes starting from 21:28:21 UT on 12.03.11. The parameters presented in this figure are deduced from the SDA measurements and consist of the total vertical electric field generated by the positive and negative lightning flashes each second together with the first and second harmonics of the Schumann resonance. The positive discharges, being in average two times less frequent with respect with the negative ones, are often associated with the stronger currents that generate the higher radiation fields. Increase of the electric field positive pulses above  $\sim 200$  mV m $^{-1}$  per second is observed to be related to an increase of the magnitude of the second ( $\sim 14.1$  Hz) harmonic of the Schumann resonance above  $\sim 60$   $\mu$ V m $^{-1}$ .

## 5. Discussion

In this Section we use the measured data set to estimate some key parameters, such as mobility and concentration of small-ions, their generation rate, as well as the radiation dose rate.

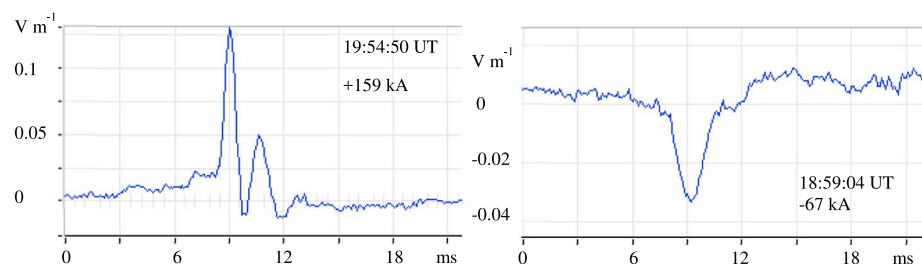
### 5.1. From Air Density to Small-ion Mobility

The mobility of small-ions is a parameter that is proportional to the ion charge and inversely proportional to the ion mass and collision frequency of the ion with other atmospheric species. Assuming that the effective mass and diameter of small ions do not change with the height, the mobility is estimated (see, for example, Bricard, 1965 or Wählin, 1994) to vary inversely with the air mass density, i.e.

$$\mu_{\pm}(z) \approx \frac{\rho_0}{\rho(z)} \mu_{\pm 0}, \quad (10)$$

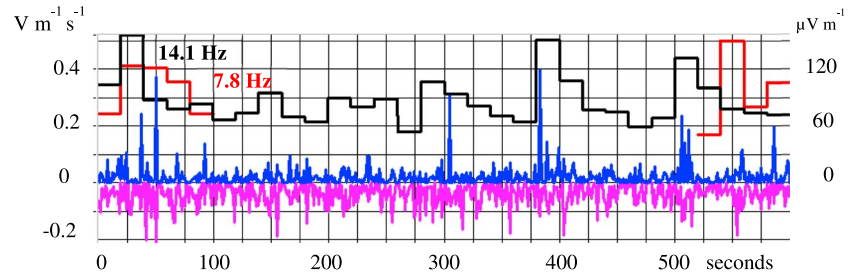
here  $\mu_{\pm 0} \approx (1.6 \pm 0.2) 10^{-4}$  m $^2$  V $^{-1}$  s $^{-1}$  at 0 °C and 10 $^3$  hPa. The negative ion mobility is quoted to be usually slightly higher than the positive one.

The air mass density, calculated in the Section 3.2.2 from the temperature and pressure measurements, is found to exponentially decrease with height as



**Figure 10.** Positive (left) and negative (right) cloud-ground strokes detected by SDA at altitude of  $\sim 19.4$  km on 15.02.2011. Strokes were produced south-west from the balloon location at the distance of  $\sim 3100$  and 3200 km, respectively.





**Figure 11.** Temporal variation of the total vertical electric field generated by the positive (bleu) and negative (magenta) lightning flashes each second together with the first (in red) and second (in black) harmonics of the Schumann resonance. The measurements are performed at the altitude of  $\sim 22$  km during 10 minutes starting from 21:28:21 UT on 12.03.11.

$$\rho(z) \sim 1.3e^{-z/9} \text{ below } z = 8 \text{ km and } \rho(z) \sim 2e^{-z/6.2} \text{ above } z = 8 \text{ km.} \quad (11)$$

Using (11), the (10) reads as

$$\mu(z) \sim 1.5 \cdot 10^{-4} e^{z/9} \text{ below } z = 8 \text{ km and } \mu(z) \sim 10^{-4} e^{z/6.2} \text{ above } z = 8 \text{ km.} \quad (12)$$

### 5.2. From Electric Conductivity of Air to Concentration of Small-ions

Considering the measured values of the electric conductivity of air (8) and estimated values of the small-ion mobility (12), the equation (7) at the altitudes above  $\sim 8$  km reads as

$$n_i(z) \approx 1.4 \cdot 10^{10} e^{-z/31.5}. \quad (13)$$

The variation of the small-ion concentration is displayed in Figure 8 together with the space charge density evaluated in Section 4.1.

### 5.3. Ion Generation Rate and Radiation Environment

One of the major sources of the atmospheric gas ionization in the Earth stratosphere is likely galactic cosmic rays and solar energetic particles. The radiation environment varies with the solar activity, geomagnetic latitude, solar zenith angle and atmospheric thickness (see, for example, Nehar, 1967; Reitz, 1993; Viggiano & Arnold, 1995; Singh et al., 2011). Usually the radiation level is measured in rad per time unit and is considered to be one of the essential parameters for the preparation of space missions, definition of the protection level of space instruments and probability of their failure.

In our example and making some assumptions, the radiation dose rate can be roughly estimated from the deduced values of the ion density (13). Assuming for a simplicity that the ion recombination coefficient,  $\alpha$ , is constant and equals  $\sim 10^{-12} \text{ m}^3 \text{ s}^{-1}$  (see, for example, Cole & Pierce, 1965 or Shreve, 1970), the ion production rate is estimated to be equal to

$$\beta(z) \approx \alpha \cdot n_i^2(z) \approx 2 \cdot 10^8 e^{-z/16} \text{ m}^3 \text{ s}^{-1}, 8 < z < 23 \text{ km.} \quad (14)$$

This expression is valid at the altitudes between 8 and 23 km, where we have an appropriate estimation of the ion density.

During its motion in a gas, each energetic particle undergoes numerous collisions with neutral species (molecules, atoms). These collisions result in a loss of their energy and in the generation of ions of both polarities. The radiation rate,  $B$  (in  $\text{rad s}^{-1}$ ), is related to the ion production rate, gas ionization energy,  $e_i$ , and gas mass density,  $\rho$ , as:

$$B \approx \frac{\beta \cdot e_i}{\kappa \rho}, \quad (15)$$

Here  $\kappa = 6.24 \cdot 10^{16} \text{ eV kg}^{-1} \text{ rad}^{-1}$  is conversion coefficient from rad to  $\text{eV kg}^{-1}$  and the gas ionization energy

is taken to be equal to  $\sim 35$  eV. Substituting the previously estimated air mass density (Figure 5) and ion production rate (14) in the equation (15), the last reads:

$$B(z) \approx 5.6 \cdot 10^{-8} e^{z/10} \text{ rad s}^{-1} = 4.8 \cdot 10^{-3} e^{z/10} \text{ rad day}^{-1}, 8 < z < 23 \text{ km.} \quad (16)$$

Thus, one gets the radiation dose rate of  $0.013 \text{ rad day}^{-1}$  at the altitude of 10 km and  $0.035 \text{ rad day}^{-1}$  at 20 km.

## 6. Conclusions

We have presented the observations performed with the SDA instrument in the auroral atmosphere during two balloon flights in February and March of 2011. Two key parameters that characterize the electric properties of the atmosphere were measured during these flights, i.e. the electric field and the electric conductivity of air. The electric field variation with the height was found to be controlled by the horizontal wind profile, while the temporal change of the electric conductivity of air was observed to correlate with the sub-storm activity. Few parameters were deduced from the data measured during the flights, such as the charge density, ion concentration, ion mobility and current density.

Lightning flashes produced in the southern Europe, i.e. more than  $\sim 3000$  km from the balloon location, were clearly detected by the SDA AC channels. An increase of the positive strokes intensity above  $\sim 200 \text{ mV m}^{-1} \text{ s}^{-1}$  was shown to result in the amplification of the Schumann resonances amplitude up to  $\sim 150 \mu\text{V m}^{-1}$ .

The radiation dose rate that produces the air ionization in the lower stratosphere was estimated to vary between  $\sim 0.01$  and  $0.05 \text{ rad day}^{-1}$ . The radiation rate of  $\sim 0.018\text{--}0.026 \text{ rad day}^{-1}$  found between 13 and 17 km is comparable with that reported by Hassler et al. [2014] and measured on the Mars surface with the Radiation Assessment Detector (RAD) on the Mars Science Laboratory's Curiosity rover in 2012–2103.

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