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# **Geophysical Research Letters**

### **RESEARCH LETTER**

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#### **Key Points:**

- An artificial neural network (ANN) is implemented to infer solar wind proxies from sheath measurements at Mars
- The ANN can determine solar wind ion density, speed, temperature, and magnetic field magnitude with high accuracies
- The ANN can determine the magnetic field cone and the clock angles with moderate accuracies

#### **Supporting Information:**

Supporting Information S1

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## An Artificial Neural Network for Inferring Solar Wind Proxies at Mars

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**Abstract** We present a novel method to determine solar wind proxies from sheath measurements at Mars. Specifically, we develop an artificial neural network (ANN) to simultaneously infer seven solar wind proxies: ion density, ion speed, ion temperature, and interplanetary magnetic field magnitude and its vector components, using spacecraft measurements of ion moments, magnetic field magnitude, magnetic field components in the sheath, and the solar extreme ultraviolet flux. The ANN was trained and tested using 3 years of data from the Mars Atmosphere and Volatile EvolutioN (MAVEN) spacecraft. When compared with MAVEN spacecraft's in situ measured values of the solar wind parameters, we find that the ANN proxies for the solar wind ion density, ion speed, ion temperature, and interplanetary magnetic field magnitude have percentage differences of 50% or less for 84.4%, 99.9%, 86.8%, and 79.8% of the instances, respectively. For the cone angle and clock angle proxies, 69.1% and 53.3% of instances, respectively, have angle differences of 30° or less.

**Plain Language Summary** We introduce a new technique for determining solar wind parameter values upstream of Mars using spacecraft-measured values of parameters in the plasma environment near Mars. This technique involves using an artificial neural network. Data in the solar wind and the plasma environment near Mars measured by the Mars Atmosphere and Volatile EvolutioN spacecraft, which has been orbiting Mars since 2014, were used to train an artificial neural network to simultaneously infer seven solar wind parameters: solar wind density, speed, temperature, and interplanetary magnetic field and its three components. Comparison of the neural network-inferred values to the in situ measured values reveals that the artificial neural network can infer the solar wind density, speed, temperature, and the interplanetary magnetic field magnitude with high accuracies and the orientation of the magnetic field with moderate accuracies. Thus, this artificial neural network can be successfully used for inferring solar wind parameters at Mars. Since Mars lacks a dedicated solar wind monitor, unlike at Earth, this technique is useful for obtaining solar wind parameters during times when a Mars orbiter does not traverse through the solar wind upstream of Mars. Knowledge of the solar wind parameter values is essential for studying how the solar wind influences Mars' atmospheric escape.

#### **1. Introduction**

#### 1.1. Need for Measuring Solar Wind Parameters

In situ observations by Phobos-2, Mars Global Surveyor, Mars Express (MEX), and recently by the Mars Atmosphere and Volatile EvolutioN (MAVEN) spacecraft have revealed that solar wind parameters, typically referred to as solar wind drivers, influence plasma boundary locations, sheath ion moments, turbulence, and atmospheric escape at Mars. Moreover, organizing observations in the Mars plasma environment based on upstream parameters enables to discover whether certain phenomena are influenced by the upstream drivers. Thus, direct measurement of solar wind parameters or deriving their proxies is crucial for investigating plasma environment dynamics at Mars. The objective of this paper is to develop a novel method of deriving solar wind proxies at Mars by using an artificial neural network (ANN) and in situ measurements of magnetosheath parameters made by the MAVEN spacecraft. Before describing our method, below, we will

©2018. American Geophysical Union. All Rights Reserved. first describe how the Mars plasma environment is influenced by the solar wind drivers to emphasize the need for direct measurement or deriving solar wind proxies at Mars.

The bow shock location at Mars is influenced by the upstream parameters such as convection electric field direction, magnetic field orientation, Mach number, dynamic pressure, and extreme ultraviolet (EUV) flux. The bow shock tends to be located at a larger distance in the hemisphere in the direction of the upstream convection electric field (Dubinin et al., 1998). The quasi-perpendicular bow shock tends to lie farther upstream than the quasi-parallel bow shock (Dubinin et al., 1998; Halekas, Brain, et al., 2017). The bow shock location lies much closer to the planet during the times of high Mach number (Edberg, Lester, et al., 2010; Halekas, Ruhunusiri, et al., 2017) and dynamic pressure (Edberg et al., 2009; Halekas, Ruhunusiri, et al., 2017; Hall et al., 2016). However, the bow shock is found to lie much farther upstream during the times of high EUV flux (Edberg et al., 2009; Halekas, Ruhunusiri, et al., 2017; Hall et al., 2016). The magnetic pileup boundary (MPB) is also influenced by the upstream parameters (Brain et al., 2005; Edberg et al., 2008, 2009).

Ion temperature and its anisotropy in the sheath and the foreshock are influenced by the upstream magnetic field orientation and Mach number (Halekas, Brain, et al., 2017). Parallel temperature has larger values in the quasi-parallel flank than in the quasi-perpendicular flank, whereas the perpendicular temperature has approximately the same values in both the quasi-parallel and quasi-perpendicular flanks. Thus, the temperature anisotropy tends to be much larger in the quasi-perpendicular flank than in the quasi-parallel flank. Under high Mach numbers, both the parallel and perpendicular temperatures have been found to increase. However, the ion temperature anisotropy has been found to be higher for low Mach numbers than for high Mach numbers.

Waves and turbulence at Mars are affected by the EUV flux, upstream convection electric field, magnetic field orientation, Mach number, and proton beta. Low-frequency waves and turbulence levels are enhanced during the times of high EUV flux (Romanelli et al., 2016; Ruhunusiri et al., 2017, 2018). The waves and turbulence show asymmetries with respect to the upstream convection electric field direction and in the quasi-parallel versus quasi-perpendicular foreshock regions and the adjacent sheath regions (Fowler et al., 2017; Ruhunusiri et al., 2017; Wei & Russell, 2006). Mach number and proton beta have also been found to influence wave occurrence rates of waves such as the 1-Hz waves (Ruhunusiri et al., 2018).

Atmospheric loss via acceleration of heavy ions is influenced by the upstream convection electric field (Dubinin et al., 2011; Johnson et al., 2018). Acceleration of ions via the convection electric field together with the large ion gyroradius effects lead to asymmetries in the heavy ion distribution in the Mars plasma environment (Dong et al., 2017; Fang et al., 2008; Jarvinen et al., 2015). Ion escape has also been found to be sensitive to the EUV flux variation at Mars (Dong et al., 2017). Ion escapes have been observed to be enhanced during upstream impulsive events such as coronal mass ejections suggesting that increase in solar wind fluxes lead to enhanced ion loss (Edberg, Nilsson, et al., 2010; Futaana et al., 2008; Jakosky et al., 2015). Unraveling how the upstream parameters influence the atmospheric escape is crucial to extrapolating Mars' atmospheric escape backward in time to determine the total atmospheric mass lost to space (Lillis et al., 2015).

Organizing the observations with the upstream parameters is necessary to determine how the Mars plasma environment phenomena are influenced by them. For example, investigators typically organize observations in the Mars Solar Electric coordinates system where *X* axis is aligned with the solar wind velocity, *Y* axis is aligned with the magnetic field component perpendicular to the solar wind velocity, and *Z* axis is aligned with the convection electric field direction. Dong et al. (2015), for example, used Mars Solar Electric coordinate system to organize ion flux observations at Mars, which enabled them to characterize the contributions from different escape channels such as tail versus plume.

#### 1.2. Previous Methods for Determining Solar Wind Proxies at Mars

Since Mars lacks a dedicated solar wind monitor, the above described discoveries were made using a derivation of solar wind proxies at Mars. We will describe three most commonly used methods below. The first method is extrapolating solar wind measurements at Earth to Mars. The second method is interpolating two successive measurements made by the orbital assets when they are located upstream of the bow shock. The third method is to use a physics-based model to infer upstream parameters using spacecraft in situ measurements of the downstream parameters. In the method of extrapolation, the solar wind parameters measured at Earth are appropriately scaled considering how these parameters vary with the heliocentric distance (Edberg, Lester, et al., 2010). For example, the solar wind density and magnetic field magnitude have a  $1/r^2$  dependence and the solar wind ion temperature has a  $1/r^{2/3}$  dependence with heliocentric distance *r*. However, this method can be utilized only when Mars and Earth are located in approximately the same solar wind sector. Comparing the extrapolated solar wind bulk velocities measurements from STEREO A and B to in situ measurements made by MEX, Opitz et al. (2010) suggested that a reliable estimation can only be made when the longitudinal separation between Earth and Mars is less than 65°. Thus, the use of this method to derive solar wind proxies at Mars is limited to typically a couple of months every 2 years.

The second method is to interpolate two adjacent measurements of a solar wind parameter made by a spacecraft when it is in the solar wind to determine the corresponding value of that parameter when the spacecraft is located downstream of the bow shock. For example, MAVEN spacecraft periodically ventures into the solar wind when its apoapsis is located outside Mars' bow shock. So when the spacecraft is located downstream of the bow shock, in order to infer the solar wind proxies, first two adjacent measurements of the solar wind parameters are linearly interpolated. Then, it is assumed that the interpolated value is equal to the value of the solar wind parameter of interest. Alternatively, one can assume that a solar wind proxy for a parameter is equal to its most recent measurement made when the spacecraft was located upstream of the bow shock. A statistical study of the solar wind parameters using the MAVEN spacecraft measurements has revealed that this latter approximation is suitable as long as the time duration is less than or equal to about a couple of hours (Marquette et al., 2018). Implementation of these techniques is not possible during the times when the spacecraft does not have upstream solar wind coverage due to a precession of the orbit.

In the third approach, certain solar wind parameters are inferred based on the spacecraft measurements of the downstream parameters using a physics-based model. For example, assuming that the magnetic field pressure in the MPB balances the incident solar wind dynamic pressure, the upstream solar wind dynamic pressure can be inferred based on magnetic field measurements in the MPB (Crider et al., 2003).

Alternatively, two spacecraft measurements can be used to obtain direct measurements of the solar wind in the absence of a dedicated monitor. When one spacecraft is located downstream of the bow shock, if a second spacecraft is simultaneously located upstream of the bow shock, the measurements of the second spacecraft in the solar wind can be used to determine the solar wind parameters corresponding to the times when the first spacecraft was located downstream of the bow shock. This method has recently been used by the MEX and MAVEN spacecraft (Harada et al., 2018). However, this ideally works if two spacecraft have identical instruments. While Mars currently has six orbiters, they all do not have identical plasma instruments. For example, while the MAVEN spacecraft has a magnetometer, the MEX lacks a magnetometer. Thus, interplanetary magnetic field (IMF) can only be obtained for times when MAVEN ventures upstream of the bow shock. Recent studies have also utilized magnetohydrodynamic models to derive solar wind proxies at Mars (Fang et al., 2018; Ma et al., 2018).

The influence of upstream parameters on the Mars plasma environment dynamics and the need for these parameters for organizing downstream observations, as we discussed in the previous section, emphasize the need for continuous solar wind monitoring or developing techniques to derive solar wind proxies that are complementary to existing methods that we discussed above. In this paper, we introduce a novel method to infer solar wind proxies at Mars. We, in particular, develop an ANN to infer solar wind proxies using MAVEN particle and field measurements in the sheath. This is the first time that an ANN has been employed to derive solar wind proxies at any planet. This is also the first time that an ANN has been used for an application involving plasma observations at Mars. The combination of our method for deriving solar wind proxies using MAVEN spacecraft observations in the sheath with its direct measurements of the solar wind parameters will enable to provide solar wind proxies for majority of times from September 2014 onward. Thus, this work should be of interest to a large scientific community who are involved in investigating the impact of the upstream drivers on the Mars plasma environment. The methods developed in this paper should also be of interest to the planetary-physics and heliospheric scientific communities.

#### 1.3. ANNs

Here we briefly introduce ANNs and their applications. ANNs consist of computational units called neurons which are analogous to the biological neurons (Samarasinghe, 2007). A neuron consists of one or more inputs and outputs. The neuron computes a weighted sum of its inputs (and also applies a bias to this weighted sum).

Then, it produces an output using a transfer function which can be either linear or nonlinear. A typical ANN contains hundreds of such neurons. The number of inputs, outputs, and transfer functions of the neurons in the ANN must be user specified. The ANN determines the weights and biases during a process called training by comparing the outputs it produces, for a set of inputs provided by the user, to a set of targets also provided by the user.

ANNs are powerful at identification of patterns in the observational data analogous to their biological counterparts. Thus, ANNs have been successfully used in a variety of applications in many fields such as hydrology (Hsu et al., 1995), chemostratigraphy (Malmgren & Nordlund, 1996), climatology (Jiang et al., 2018), seismology (Dai & MacBeth, 1997), and cosmology (George & Huerta, 2018). Applications of ANNs in space physics include Earth's magnetospheric response and parameter prediction (Chu, Bortnik, Li, Ma, Angelopoulos, & Thorne, 2017; Chu, Bortnik, Li, Ma, Denton, et al. 2017; Gleisner & Lundstedt, 1997), ionospheric parameter prediction (Tulunay et al., 2006), and plasma wave identification (Ruhunusiri, 2018). In this paper, we will describe the development of an ANN to derive solar wind proxies at Mars.

### 2. Method

Our overall approach was to develop an ANN that can simultaneously infer seven solar wind proxies: solar wind ion density, ion speed, ion temperature, and IMF magnitude and its three vector components. As we discussed in section 1.1, there is an interrelation between sheath ion moments and magnetic field to the solar wind ion moments, IMF, and EUV flux, and these interrelations also generally depend on the spatial location within the Mars plasma environment. Thus, we used following inputs to the ANN: magnetosheath location, magnetosheath ion density, ion speed, ion temperature, magnetic field magnitude, magnetic field components, and EUV flux. Since the dependencies of the sheath parameters to the solar wind parameters may not be necessarily linear, we used nonlinear elements in the ANN such as nonlinear transfer functions.

#### 2.1. ANN Architecture

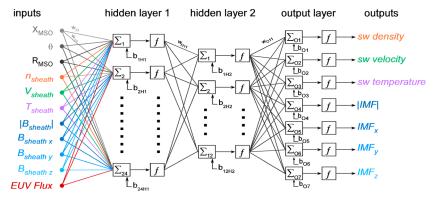
We used a feedforward neural network (Beale et al., 2010) with 11 inputs, a normalization layer, two hidden layers, an output layer, a denormalization layer, and seven outputs (see Figure 1). Similar neural network architectures have been successfully used in nonlinear regression applications (Rojas, 1996). The inputs are the spacecraft location in cylindrical coordinates ( $X_{MSO}$ ,  $R_{MSO}$ , and  $\theta = \tan^{-1}(Y_{MSO}/Z_{MSO})$  where MSO stands for Mars Solar Orbital coordinate system), the magnetosheath ion density, ion speed, ion temperature, magnetic field magnitude, magnetic field components (in MSO coordinates), and the EUV flux. A normalization layer was used to normalize the input values between -1 and +1. The first hidden layer contains 24 neurons each with a hyperbolic tangent sigmoid (tansig) transfer function which is a standard nonlinear function used in neural networks, and this has the form  $2/(1 + e^{-2n}) - 1$  where  $n = \sum (I_i \cdot w_i + b_i)$ . Here  $I_i$  are the input values,  $w_i$ are the weights, and  $b_i$  are the bias values. The reason for normalization of the inputs to the range between -1 and 1 is that the output of the tansig transfer function reaches limiting values of -1 and 1 for *n* values less than -1 and greater than 1, respectively. The second hidden layer contains 12 neurons each also with tansig transfer functions. The output layer contains seven neurons each with a linear transfer function of the form y = x where y is the output of a neuron and x is the input to a neuron. The outputs from the output layer are then denormalized by the denormalization layer to scale the parameters to the realistic range of values. Finally, the ANN produces seven outputs: solar wind ion density, ion speed, ion temperature, IMF magnitude, and IMF components.

The ANN for solar wind inference was constructed, trained, validated, and tested using the MATLAB neural network toolbox (Beale et al., 2010; Demuth & Beale, 2002). We will describe the data selection, ANN training, validation, and testing procedures below. Training and validation were used to train the network and to select the best architecture. Testing was used to determine the ability of the network to infer the solar wind parameters from sheath measurements.

#### 2.2. Data Selection

To perform ANN training, validation, and testing, we require sheath parameters and solar wind parameters corresponding to the times where the sheath parameter measurements were made so that during training the ANN can learn the relationships between the sheath parameters and the solar wind parameters. We selected 3 years of MAVEN spacecraft data spanning from 1 December 2014 to 31 January 2018 to perform ANN training, validation, and testing. Whether the spacecraft was in the solar wind was determined using an automated procedure described in section 3.1 of Halekas, Ruhunusiri, et al. (2017). When the spacecraft was determined





**Figure 1.** The architecture of the ANN used for inferring solar wind proxies using sheath measurements. The ANN has 11 inputs for spacecraft location, sheath ion moments, magnetic field magnitude and components, and EUV flux and seven outputs for solar wind ion moments and magnetic field magnitude and components. The ANN consists of two hidden layers consisting of 24 and 12 neurons in the first and second hidden layers, respectively. The neurons in the hidden layers have tansig nonlinear transfer functions, and the neurons in the output layer have linear transfer functions. ANN = artificial neural network; EUV = extreme ultraviolet; MSO = Mars Solar Orbital coordinate system; sw = solar wind; IMF = interplanetary magnetic field.

to be in the solar wind, an orbit average of the solar wind parameters was determined for each orbit. We used the Solar Wind Ion Analyzer instrument (Halekas, Ruhunusiri, et al., 2017; Halekas et al., 2015) measurements to determine the solar wind ion moments (density, speed, and temperature) and the magnetometer (MAG) (Connerney, Espley, Lawton et al. 2015, Connerney, Espley, DiBraccio et al. 2015) to obtain the IMF components. The distribution of the measured solar wind ion moments and IMF magnitude and orientations (clock and cone angles) are depicted in Figure 2. Time segments where the spacecraft was traversing through the sheath was identified by determining whether the spacecraft was located between the nominal bow shock and the MPB using the boundary models by Trotignon et al. (2006). Then, 4-min averages of the sheath ion moments (determined using the Solar Wind Ion Analyzer instrument) and magnetic field magnitudes and its components (determined using the MAG instrument) were computed to obtain average sheath parameters to minimize the effects of turbulent fluctuations in the sheath. Then, for each of the locations in the sheath, we determined the corresponding solar wind parameters by assuming that they are equal to the value of the temporally closest measurements of the two adjacent solar wind measurements. We required that the temporally closest measurement of the solar wind parameter should be available within a temporal separation of less than 2 hr. Otherwise, both the sheath parameters and the corresponding solar wind parameters were removed from the data set. We used the Lyman-alpha flux measured by the EUV monitor (Eparvier et al., 2015) on board the MAVEN spacecraft as an input to the ANN which is a proxy for the EUV flux.

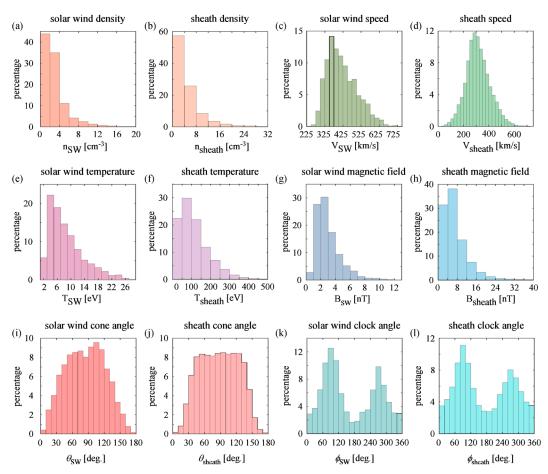
#### 2.3. ANN Training, Validation, and Testing

We used 80% of the data for training and validation and the remaining 20% of the data for testing as typically done for ANNs after randomizing the data. The sheath parameters were used as the inputs to the ANN, while the solar wind parameters were used as targets. The ANN was trained using the resilient backpropagation algorithm (Riedmiller & Braun, 1993) which determines the weights and biases of the neurons minimizing the difference between network outputs and targets. We selected this algorithm because it yielded fast convergence times while producing high accuracies.

A tenfold cross validation (Setiono, 2001) was performed using the training and validation data set to obtain the best ANN architecture containing 24 neurons in the first hidden layer and 12 neurons in the second hidden layer that we discussed in section 2.1. When the network complexity is lower, for example, ANNs consisting of lower than 24 neurons in the first hidden layer and less than 12 neurons in the second layer, the accuracies are lower. For more complex architectures, for example, ANNs consisting of more than 24 neurons in the first hidden layer and more than 12 neurons in the second hidden layer, the network accuracies do not significantly improve compared to the one with 24 and 12 neurons in the first and second hidden layers. Also, as the complexity of the ANN architecture is increased, the network training time increases and the ANN tends to memorize the relationships between the inputs and targets rather than generalizing the relationships between them (Samarasinghe, 2007). Thus, we selected the ANN with 24 neurons in the first hidden



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**Figure 2.** (a, c, e, g, i, k) Distributions of the MAVEN measurements of the solar wind ion moments and IMF magnitude and orientations. (b, d, f, h, j, l) Distributions of the MAVEN measurements of sheath ion moments and magnetic field magnitude and orientations. Here  $n_{SW}$ ,  $V_{SW}$ ,  $T_{SW}$ ,  $B_{SW}$ ,  $\theta_{SW}$ , and  $\phi_{SW}$  denote solar wind ion density, ion speed, ion temperature, IMF magnitude, IMF cone angle, and IMF clock angle, respectively. These data were used for training, validation, and testing the artificial neural network. Note that different scales are used for plotting the solar wind and sheath ion moments and IMF magnitude. Here cone angle  $= \cos^{-1}(B_x/|B|)$  and clock angle  $= \tan^{-1}(B_y/B_z)$ , where  $B_{x}$ ,  $B_{y}$ , and  $B_z$  are the sheath or IMF magnetic field components in Mars Solar Orbital coordinates and |B| is the sheath or IMF magnetic field magnitude. We use the four quadrants in the  $B_y$ - $B_z$  plane determined by the signs of  $B_y$  and  $B_z$  to obtain 0° – 360° value range for the clock angle. MAVEN = Mars Atmosphere and Volatile EvolutioN; IMF = interplanetary magnetic field.

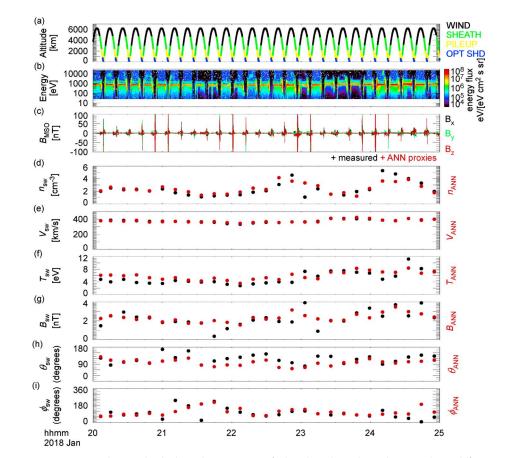
layer and 12 neurons in the second hidden layer as the optimum ANN architecture for deriving solar wind proxies using sheath measurements.

After selection of the optimum network architecture, the testing data set was used to assess the accuracies of the ANN. A comparison of the measured solar wind parameters and ANN-inferred solar wind proxies for a 5-day period in the testing data set is depicted in Figure 3. The ANN inference accuracies and uncertainties associated with those inferences, determined using the entire testing data set, are depicted in Figure 4 (see Figure S1 in the supporting information for the corresponding parameters for an ANN trained and tested using 70% and 30% of data, respectively). The uncertainties shown in Figure 4 are the 50% and 75% confidence intervals for the absolute difference between the ANN estimates based on the 4-min averaged sheath values and the corresponding orbit-averaged values of the solar wind parameters.

### 3. Results and Discussions

We find that the ANN can infer the solar wind ion density, ion speed, ion temperature, and magnetic field magnitude with high accuracies and the clock angle and the cone angle with moderate accuracies. Figure 3 depicts a comparison of the measured solar wind parameters and the ANN solar wind proxies for a 5-day





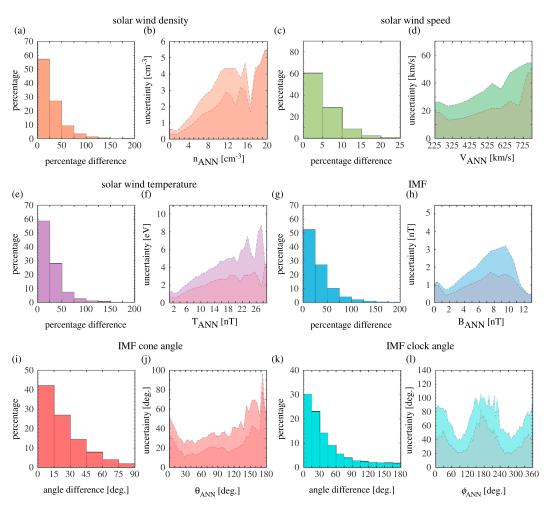
**Figure 3.** (a) Mars Atmosphere and Volatile EvolutioN spacecraft altitude with a color coding to indicate different regions in the Mars plasma environment. (b) A times stack of ion energy spectra from Solar Wind Ion Analyzer instrument which were used to determine the solar wind and sheath ion moments. (c) Time series of the three components of the magnetic field in Mars Solar Orbital coordinate system measured by the MAG instrument. (d–i) Measured versus ANN proxies for solar wind ion moments and IMF magnitude and orientations. The measured solar wind values are depicted in black, whereas the ANN proxies are depicted in red. Here  $n_{ANN}$ ,  $V_{ANN}$ ,  $T_{ANN}$ ,  $B_{ANN}$ ,  $\theta_{ANN}$ , and  $\phi_{ANN}$  denote ANN proxies for solar wind ion density, ion speed, ion temperature, IMF magnitude, IMF cone angle, and IMF clock angle, respectively. Each data point represents the orbit averages of the measured solar wind values and the ANN proxy values. ANN = artificial neural network; IMF = interplanetary magnetic field.

period. Note that Figure 3 depicts the orbit-average values of the measured upstream values and the ANN predicted values. For the time period shown in Figure 3, for a majority of times, the ANN solar wind proxies lie within a factor of 2 of the solar wind measurements.

In order to assess the overall performance of the ANN, we computed the accuracies of the ANN solar wind proxies using the entire testing data set. For ion moment and the magnetic field magnitude ANN proxies, we computed the percentage differences as a measure of their accuracies where percentage difference = |ANN proxy-measured value| × 100/(measured value), while for the magnetic field cone and clock angle ANN proxies, we computed the difference between the ANN proxy values and the measured values as a measure of the angle accuracies. The histograms of these percentage differences and angle differences for the ANN solar wind proxies are shown in Figure 4. The percentage of instances where the ANN proxies for the solar wind ion density, ion speed, ion temperature, and IMF magnitude have percentage differences of 50% or less are 84.4%, 99.9%, 86.8%, and 79.8%, respectively. For the cone angle and clock angle proxies, 69.1% and 53.3% of instances, respectively, have angle differences of 30° or less. Thus, accuracies are highest for the ANN solar wind ion speed proxies followed by ion density, temperature, and IMF magnitude proxies. Accuracies for ANN cone angle proxies are higher than those for the clock angle proxies.

To determine the uncertainties associated with the ANN proxies, we plot confidence interval plots in Figure 4. The *X* axis in these confidence interval plots depict the ANN proxy value, and the two shaded regions show





**Figure 4.** (a–h) Histograms for the percentage differences and confidence interval plots of uncertainties for the ANN solar wind ion moment and IMF magnitude proxies. (i–I) Histograms for angle differences and confidence interval plots of uncertainties for the ANN IMF cone and clock angle proxies. The two shaded regions in panels (b), (d), (f), (h), (j), and (l) depict the 50% and 75% confidence intervals for the uncertainties. The ANN has very good accuracies for solar wind ion moment and IMF magnitude inferences, whereas it has moderate accuracies for IMF cone and clock angle inferences. ANN = artificial neural network; IMF = interplanetary magnetic field.

where the uncertainties have 50% and 75% probabilities of being less than the upper boundary value of these regions. For example, if the ANN solar wind ion density proxy value is 6 cm<sup>-3</sup>, there is a 50% probability that the uncertainty is less than 1.1 cm<sup>-3</sup> and a 75% probability that the uncertainty is less than 2.1 cm<sup>-3</sup> (see Figure 4b).

The ANN proxies have small uncertainties for values where they occur with a high percentage in the solar wind. For example, 94% of the time, the solar wind ion density is less than 8 cm<sup>-3</sup> and the remaining 6% of the times it has values greater than 8 cm<sup>-3</sup> as can be seen in Figure 2a and the ANN solar wind ion density proxy values have uncertainties of typically 1 cm<sup>-3</sup> for values less than 8 cm<sup>-3</sup> (here we determine the typical uncertainty using the 75% confidence interval for the uncertainty shown in Figure 4b), whereas the uncertainties are larger, typically 3.5 cm<sup>-3</sup>, when the ANN proxy value is above 8 cm<sup>-3</sup>. If we consider another example, the uncertainties for the ANN proxies for the IMF clock angles are small near 90° and 270° as can be seen in Figure 4l. As can be seen in Figure 2k, these clock angles occur with a high percentage in the solar wind. Thus, the larger uncertainty associated with values that occur with a lower probability in the solar wind should be a consequence of the ANN not being able to sufficiently learn the relationships between the sheath parameters and the solar wind parameters for those range of values. However, as the MAVEN mission progresses, we will have more data to train the ANN. So the network performance can be improved even further with the use of more data.

The sheath parameters are sensitive to variability on suborbit timescales to the upstream parameters, and this is not captured in the training data set as we used orbit-averaged values for the upstream parameters to train and test the ANN. Consequently, the uncertainties shown in Figure 4 do not include the uncertainties arising from the suborbit variation of the upstream parameters. However, if orbit-averaged ANN inferred values are used as shown in Figure 3 instead of the ANN-inferred values based on the 4-min averaged sheath values, the uncertainties shown in Figure 4 can be considered as worst case limits (see Figure S2 in the supporting information).

The reader may be curious as to why the magnetic field magnitude and its three components were used as inputs to the ANN instead of merely the three components of the magnetic field or the magnetic field magnitude, cone angle, and the clock angle. The reason for this is that, during preliminary training and testing of ANNs using the latter mentioned input combinations, we found that the ANN-inferred accuracies for the IMF magnitude, clock angle, and the cone angle are much lower than those of the ANN that used the magnetic field magnitude and its three components as inputs (see Figure S3 in the supporting information).

### 4. Summary

We developed an ANN to infer solar wind proxies using sheath measurements. The ANN was trained, validated, and tested using MAVEN data spanning 3 years. The ANN can simultaneously infer seven solar wind parameters: ion density, ion speed, ion temperature, and magnetic field magnitude and its three components using the corresponding measurements in the sheath and the EUV flux. We find that the ANN has very good accuracies for the ion moment and magnetic field magnitude inferences and moderate accuracies for magnetic field orientation inferences. We also find that the ANN proxies have small uncertainties for values that occur with larger probability in the solar wind than those value that occur with lower probability. Combining the ANN proxies of the solar wind with the measured values by the MAVEN spacecraft enables to provide a solar wind data set at Mars for a majority of times from September 2014 onward which should be invaluable to numerous investigators who are involved in studying the response of the Mars plasma environment to the solar wind drivers. The techniques developed here should also be of interest to the planetary physics and heliospheric scientific communities.

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